



Weight and Balance

N Number: _____ Date: _____

Max Ramp WT: _____ Max T/O WT: _____

	Weight	×	Arm	=	Moment
Basic Empty Weight					
Pilot & Passenger	+				
Rear Passengers	+				
Baggage	+				
Zero Fuel Weight	=				
Fuel _____ Gal	+				
Ramp Weight (Cannot Exceed Max Ramp WT)	=				
Start/Taxi/Run-Up	-				
T/O Weight (Cannot Exceed Max T/O WT)	=				
CG Within Limits? <input type="checkbox"/> Yes <input type="checkbox"/> No					
Expected Fuel Burn	-				
Expected Landing Weight	=				
CG Within Limits? <input type="checkbox"/> Yes <input type="checkbox"/> No					
Minimum fuel required for proposed flight: _____ Gal					



Performance

	GROUND ROLL	50 FT CLEARANCE
TAKEOFF		
LANDING		

Pressure Altitude: _____

Density Altitude: _____

Active Runways/Length: _____

Headwind Component: _____

Crosswind Component: _____

 PA = ((29.92 - Current ALT Setting) × 1000) + Field Elevation
 Density Altitude = Pressure Alt + (120x(OAT-ISA))

V-SPEEDS